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Vehicle Configuration (4L40H+S139)+GS2 (GL40HT)+ CUS15+4m OPLF GSLV-F12 is the 15<sup>th</sup> flight of India's Geosynchronous Satellite Launch Vehicle (GSLV) and the 9<sup>th</sup> flight with Indigenous Cryo stage. This is the 6<sup>th</sup> operational flight of GSLV with indigenous Cryogenic stage. The configuration of GSLV-F12 Payload Fairing is 4 m dia Ogive version. The GSLV-F12 will place NVS-01 satellite into a Geosynchronous Transfer Orbit. Launch is planned from the Second Launch Pad (SLP) at Satish Dhawan Space Centre, SHAR on 29<sup>th</sup> May 2023.

### **GSLV-F12 Vehicle Characteristics**

Vehicle Height	51.7 m			
Lift off Mass	420 t			
Stages	3			
First Stage (GS1)	S139+4L40H			
Second Stage (GS2)	GL40HT			
Third Stage (GS3)	CUS 15 (Indigenous)			

### **GSLV-F12 Stage Characteristics**

First Stage	First Stage (GS1)		Third Stage (GS3)
4 L40H	S139		
19.682	20.176	11.958	9.894
2.1	2.8	2.8	2.8
UH25 & N <sub>2</sub> O <sub>4</sub>	НТРВ	UH25 & N <sub>2</sub> O <sub>4</sub>	LH <sub>2</sub> & LOX
170.688	138.102	42.098	14.420
190.688	160.807	47.246	17.051
	4 L40H 19.682 2.1 UH25 & N <sub>2</sub> O <sub>4</sub> 170.688	4 L40H S139   19.682 20.176   2.1 2.8   UH25 & N₂O₄ HTPB   170.688 138.102	4 L40H S139   19.682 20.176 11.958   2.1 2.8 2.8   UH25 & N2O4 HTPB UH25 & N2O4   170.688 138.102 42.098

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# GSLV-F12/NVS-01 MISSION

### **GSLV-F12 Mission Specifications**

Orbit	GTO
Perigee	170 ± 3 km
Apogee	36568 km
Argument of Perigee	178 ± 0.5 degree
Inclination	19.36 ± 0.1 degree
Launch Azimuth	104 degree
Payload Mass	2232 kg

### **GSLV-F12 Flight Sequence**

### **Highlights in GSLV-F12**

- Sixth Operational Flight with Indigenous Cryostage
- Second Mission with 4m dia **Ogive PLF**
- Active Pressurization for CUS
- Induction of CUSCE V2



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# GSLV-F12 Flight Events

Event	Time (s)	Altitude (km)	Relative Velocity (m/s)	Inertial Velocity (m/s)
4L40Hs ignition	-4.8	0.03	0	451.92
S139 ignition	0	0.03	0	451.92
4L40H shut off	148.9	69.53	2283.02	2706.35
GS2 ignition	149.5	69.97	2283.93	2707.45
GS1 /GS2 separation	151.1	71.14	2282.18	2706.43
IS <sup>1</sup> / <sub>2</sub> M separation	156.9	75.25	2329.71	2756.48
PLF separation	237.4	114.76	3437.58	3882.17
GS2 shut off	291.3	130.38	4761.32	5207.93
GS2 separation	294.7	131.32	4775.16	5221.59
CUS ignition	295.7	131.59	4774.64	5221.11
CUS shut off	1105.2	236.67	9760.63	10212.43
CUS burn out	1110.2	241.45	9759.92	10211.88
NVS-01 S/C separation	1120.2	251.52	9750.69	10203.03





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# GSLV-F12/NVS-01 MISSION

### Navigation with Indian Constellation (NavIC)

The Indian Regional Navigation Satellite System (IRNSS) with an operational name of NavIC stands for Navigation with Indian Constellation. It is providing accurate real-time positioning and timing services over India and a region extending approximately 1500 km around the Indian Mainland. The fully deployed NavIC constellation consists of seven satellites in GeoSynchronous / Inclined GeoSynchronous Orbits.

## NVS-01 satellite



both passive and active thermal management, unified bi-propellant propulsion system and three-axis body stabilised zero momentum system with reaction wheels. The mission life of NVS-01 is expected to be better than 12 years.

As compared to the first generation satellite series, the second-generation satellite series includes L1 Navigation band and encompasses indigenously developed Rubidium atomic clock. The L1 navigation band is popular for providing PNT (Position, Navigation and Timing) services for civilian users and for interoperability with other GNSS signals. The space-qualified Rubidium atomic clock, indigenously developed by Space Applications Centre, Ahmedabad is an important technology which only a handful of countries possess.

NVS-01 is the first in the second-generation navigation satellite series which will ensure continuity of legacy NavIC services and also provide new service in L1 band. NVS-01 satellite is built around the standard I-2K Bus and is compatible with GSLV. It has a lift-off mass of approximately 2232 kg. It carries navigation payloads operating in L1, L5 and S bands. The satellite is powered by two solar arrays, capable of generating power up to 2.4 kW and a Lithium-ion battery supporting payload and bus load during eclipse. The satellite employs





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### PAYLOADS

#### **Navigation Payloads**

The navigation payload operates in L1, L5, and S bands and employs a Tri-band antenna. The heart of the navigation payload is the Rubidium Atomic Frequency Standard (RAFS), the atomic clock which acts as a stable frequency reference for the navigation payload.

#### **Ranging Payload**

The ranging payload consists of CxC transponder used for two-way CDMA ranging to facilitate precise orbit determination.

#### **Applications of NavIC:**

The key applications of NavIC include the following:

- Terrestrial, aerial, and maritime navigation
- Precision agriculture
- Geodetic surveying
- Emergency services
- Fleet management
- Location-based services in mobile devices
- Orbit determination for satellites
- Marine fisheries
- Timing services for financial institutions, power grids, and other government agencies
- Internet-of-Things (IoT) based applications
- Strategic applications





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# GLIMPSES



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Capacity Building and Public Outreach (CBPO)

Indian Space Research Organisation Department of Space, Government of India Antariksh Bhavan, New BEL Road Bengaluru-560 094, India Telephone : +91 80 22172119



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